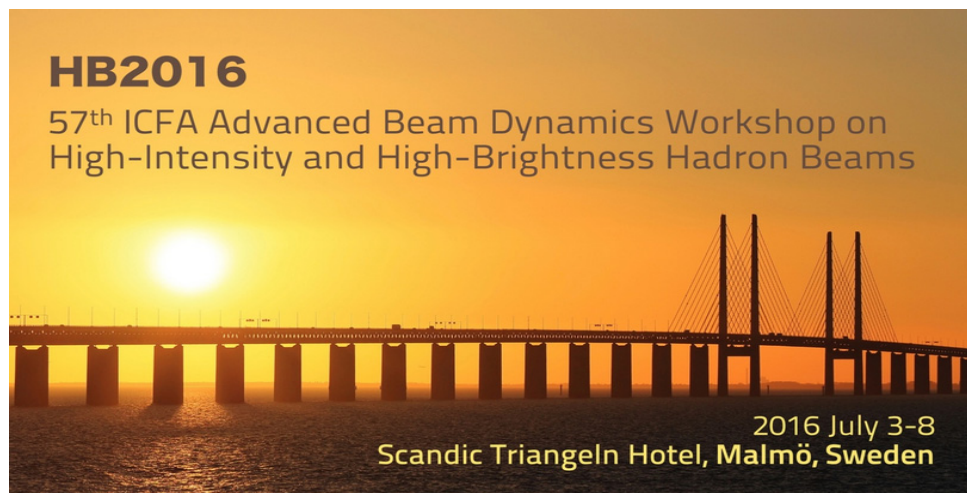




# APPLICATION OF THE OPTIMIZATION ALGORITHM IN THE COLLIMATION SYSTEM FOR CSNS/RCS

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# Outline



1. Introduction
2. Physics analysis and modeling
3. Optimization
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## 1. Introduction

1.1 Optimization and applications

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1.3 Motivation of our work

2. Physics analysis and modeling

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# Introduction-*optimization and applications*



➤ A particle accelerator is typically a complex system that consists of many components. There are many variables to be tuned to achieve good machine performance. Therefore, optimization usually needs to be performed **in a multi-dimensional variable space**.

➤ **In the design phase**, optimization and applications:

- Programs: MAD, SAD, AT, ORBIT, ...
- Algorithms: Genetic algorithm (GA), Particle swarm optimization (PSO), ...

**Single-objective** optimization methods  
GA (1975); PSO (1995)

**Multi-objective** optimization methods  
MOGAs; MOPSO

**Parallel** optimization techniques  
PGA; PPSO

- Have been developed to model the system and to further optimize the expected machine performance.

# Introduction-*optimization and applications*



- It is noted that, along with the unceasing development of computer technology and optimization algorithms, **online optimization of accelerator performance** has become both **imperative** and **feasible**.
- Several algorithms have been proposed **for different purposes** in many laboratories and have been explored for their application to **online optimization problems**;
  - Response matrix method **APS, ..., BEPCII** **【Y. Chung, G. Decker, K. Evans, 10.1109/PAC.1993.309289】**  
**【D. H. Ji, Q. Qin, et al., Proceedings of IPAC 2010, p.1644】**
  - The simplex method **KEKB** **【Y. Funakoshi, et al., EPAC2008, Genoa, Italy, 2008, p.1893】**
  - Random Walk **SLS** **【M. Aiba, et al., IPAC2012, New Orleans, Louisiana, USA, TUPPC033】**
  - Genetic Algorithm **SLAC** **【K. Tian, J. Safranek, Y. Yan. Phys. Rev. ST Accel. Beams, 2014, 17: 020703】**
  - Robust Conjugate Direction Search method **SLAC, BEPCII**  
**【X. Huang, J. Corbett, J. Safranek, J. Wu, NIM A 726 (2013) 77-83】**  
**【X. B. Huang, J. Safranek, Phys. Rev. ST Accel. Beams, 18: 084001 (2015)】**  
**【H. F. Ji, Y. Jiao, S. Wang, et al., Chinese Physics C Vol. 39, No. 12 (2015) 127006】**
  - .....

# Introduction-*the RCDS method*



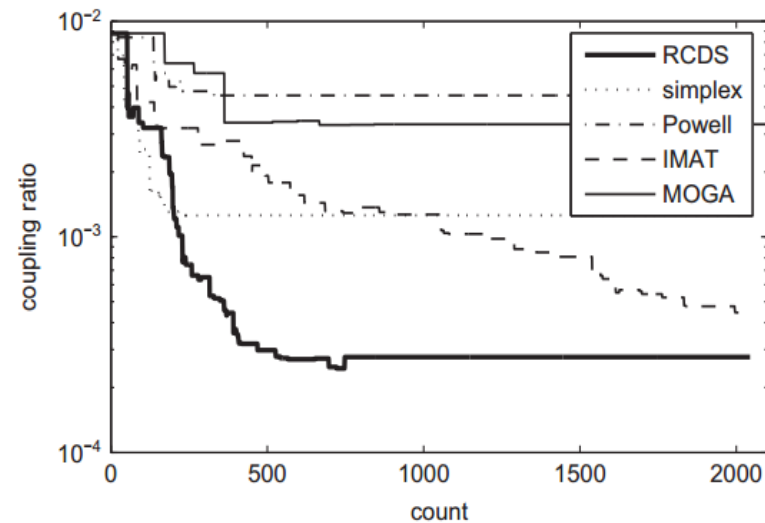
## ➤ The Robust Conjugate Direction Search method, RCDS

- The RCDS method combines the conjugate direction set approach of **Powell's method** with **a robust line optimizer** that is robust against random noise and outliers. Simulation and experimental studies have been carried out to demonstrate the strength of the algorithm.

- ✓ Powell + robust line optimizer
- ✓ Search on a conjugate direction set
- ✓ have high tolerance to noise

✓ Find a local minimum

✓ Be effective in optimizing a single-objective function of several variables



【X. Huang, J. Corbett, J. Safranek, J. Wu, *NIM A* 726 (2013) 77-83】

# Introduction-*Motivation of our work*



- The RCDS method is **much more robust against noise** than traditional algorithms. Based on the success of the RCDS method with a light source and a collider, it is interesting to **find possible applications** of this method in optimizing the performance of the collimation system for CSNS/RCS.
- To improve the efficiency of the optimization in the application, **the parallel technique** of the RCDS method is explored.
- Try to implement the RCDS method to optimize the collimation system of CSNS/RCS, and **reduce the uncontrolled beam loss**.
- **To present a way** to find an optimal parameter combination of the secondary collimators for a machine model in preparation for CSNS/RCS commissioning.



1. Introduction
2. Physics analysis and modeling
  - 2.1 Physics variables
  - 2.2 Objective function
  - 2.3 Implementation of the simulation code
  - 2.4 Parallel computing of ORBIT instances
3. Optimization
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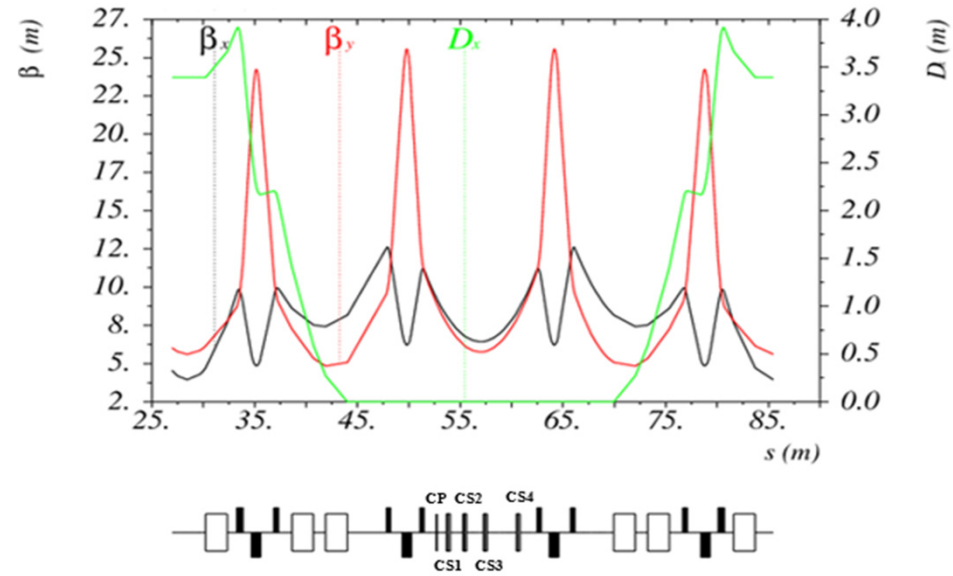


# The transverse collimation system for CSNS/RCS



- To meet the requirements for hands-on and safe maintenance, a two-stage collimation system was designed to localize the beam loss in the collimation section in the RCS.
- The transverse collimation system consists of one primary collimator and four secondary collimators.

Parameters	Symbol, unit	Value
Circumference	$C$ , m	227.92
Injection energy	$E_{inj}$ , MeV	80
Extraction energy	$E_{ext}$ , GeV	1.6
Betatron tune (H/V)	$\nu_x/\nu_y$	4.86/4.78
Accumulated particles per bunch	$N_p \times 10^{12}$	7.8
Harmonic number	$h$	2
Repetition frequency	$f_0$ , Hz	25
Accumulated and accelerated time per cycle	$t$ , ms	~20
Transverse acceptance	$\varepsilon$ , $\pi\text{mm}\cdot\text{mrad}$	540

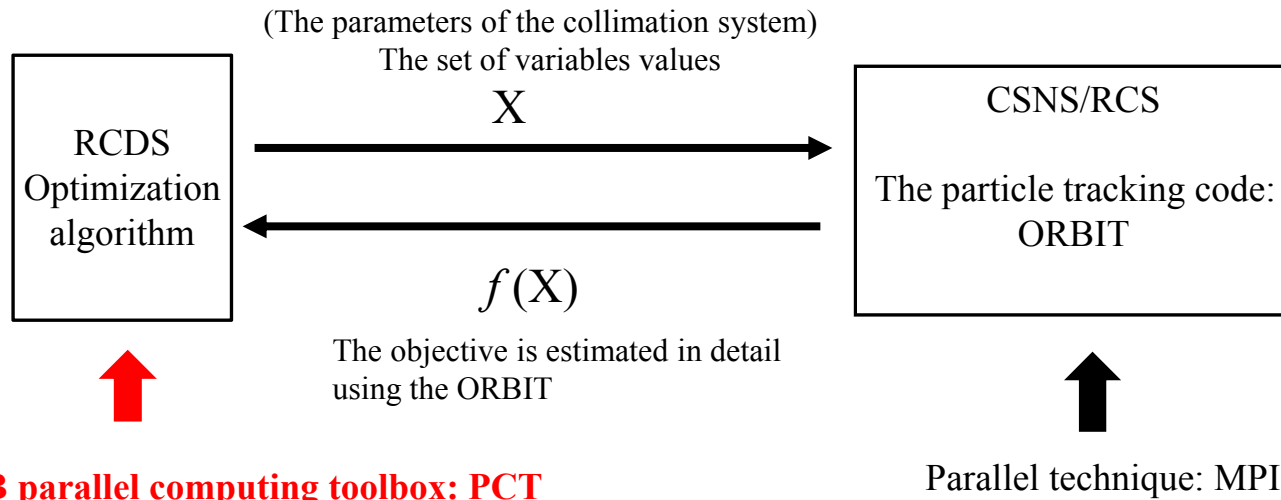


- The primary collimator has minimum acceptance in the ring, and thus is used to scatter the protons with large deviations from the beam center.
- The scattered protons are then absorbed by four secondary collimators located downstream of the primary collimator.

# The layout of the application



- In the RCS, the aperture of each secondary collimator can be varied by adjusting the positions of **four movable blocks**. To achieve satisfactory machine performance, it is necessary to **optimize the parameters of collimators**, so as to absorb most of the undesirable protons by the collimation system and minimize the beam loss in other regions of the ring.
  - In the study, we introduce an algorithm, **the RCDS method**, in the optimization.
  - The collimation process in the presence of space charge is simulated with **the Objective Ring Beam Injection and Tracking (ORBIT) code**.



# Physics variables



- The beam power is 100 kW; **the acceptance of the primary collimator is fixed to 350  $\pi$ mm·mrad all the time**, and **the secondary collimators are tuned** to optimize the performance of the collimation system.
- Each of the secondary collimators is composed of four movable copper blocks. Each block has a circular surface based on the equation:

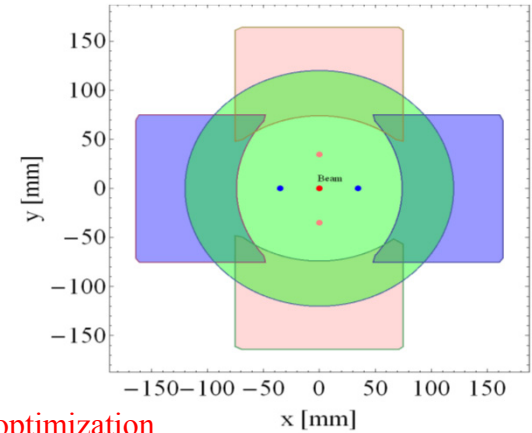
$$\frac{\left(x \cos\left(\frac{\pi\theta}{180^\circ}\right) + y \sin\left(\frac{\pi\theta}{180^\circ}\right) \pm c\right)^2}{a^2} + \frac{\left(-x \sin\left(\frac{\pi\theta}{180^\circ}\right) + y \cos\left(\frac{\pi\theta}{180^\circ}\right)\right)^2}{a^2} = 1,$$

$$\theta = \begin{cases} 0^\circ, & \text{The horizontal blocks} \\ 90^\circ, & \text{The vertical blocks} \end{cases}$$

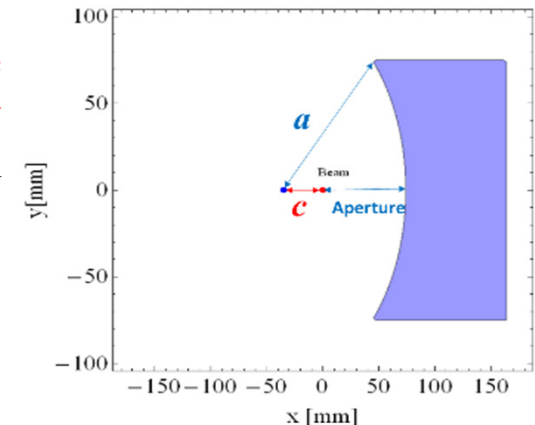
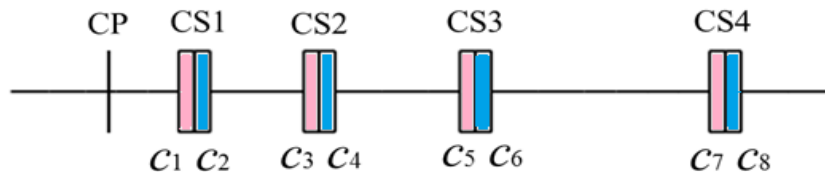
$a = 108.8$  mm  
Aperture  $\rightarrow$  (40 mm, 74 mm)

$c \rightarrow$  (34.8 mm, 68.8 mm)

The parameter  $c$  is selected to be the variable for the optimization.



- Considering the symmetry of beam distribution in simulations, **the parameters of the blocks on the same direction of each secondary collimator are the same**. Then there are **eight variables** to be tuned for four secondary collimators



# Objective function



- A single objective function is constructed to measure the performance of the collimation system:

$$f = -\eta_{\text{system}} \cdot R_{\epsilon_{xm}} \cdot R_{\epsilon_{ym}} \cdot R_{\epsilon_{add}} \cdot R_{\epsilon_{flag}}$$

- A minus sign is added to form a minimization problem.
- The cleaning efficiency of the system:

$$\eta_{\text{system}} = \frac{N_{\text{incol}} + N_{\text{indrift}}}{N_{\text{loss}}},$$

- The weight factors:

$$R_{\epsilon_{xm}} = \begin{cases} 1, & \text{if } \epsilon_{xm} \leq \text{Ref\_collimator}; \\ \text{Ref\_collimator} / \epsilon_{xm}, & \text{if } \epsilon_{xm} > \text{Ref\_collimator}. \end{cases}$$

$\epsilon_{xm}$  is the maximum 99% horizontal emittance of the beam  
 $\epsilon_{ym}$  is the maximum 99% vertical emittance of the beam

$$R_{\epsilon_{ym}} = \begin{cases} 1, & \text{if } \epsilon_{ym} \leq \text{Ref\_collimator}; \\ \text{Ref\_collimator} / \epsilon_{ym}, & \text{if } \epsilon_{ym} > \text{Ref\_collimator}. \end{cases}$$

$\text{Ref\_collimator} = 350 \pi \text{mm} \cdot \text{mrad}$

$$R_{\epsilon_{add}} = \begin{cases} 1, & \text{if } \epsilon_{add} \leq \text{Ref\_add}; \\ \text{Ref\_add} / \epsilon_{add}, & \text{if } \epsilon_{add} > \text{Ref\_add}. \end{cases}$$

$\text{Ref\_add} = 700 \pi \text{mm} \cdot \text{mrad}$   
 $\epsilon_{add} = \epsilon_{xm} + \epsilon_{ym}$

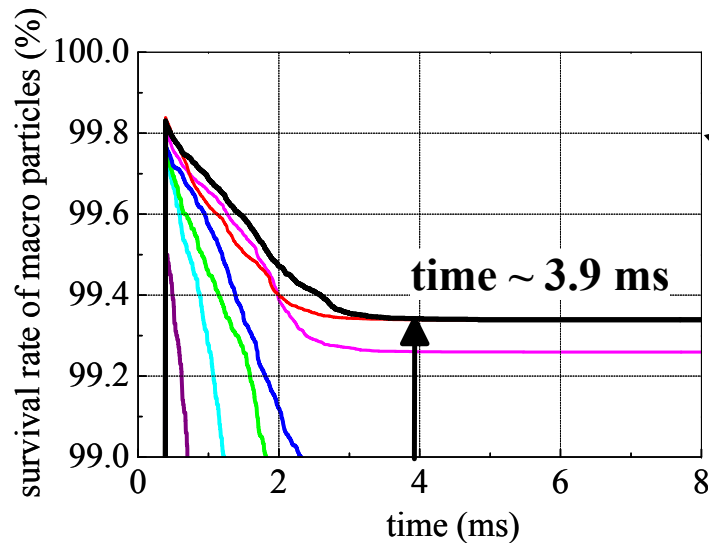
$$R_{\epsilon_{flag}} = \begin{cases} 1, & \text{if } \epsilon_{flag} \leq \text{Ref\_flag}; \\ \text{Ref\_flag} / \epsilon_{flag}, & \text{if } \epsilon_{flag} > \text{Ref\_flag}. \end{cases}$$

$\epsilon_{flag} = \max\left(\frac{\epsilon_{xm}}{\epsilon_{ym}}, \frac{\epsilon_{ym}}{\epsilon_{xm}}\right) - 1$   
 $\text{Ref\_flag} = 0.5$

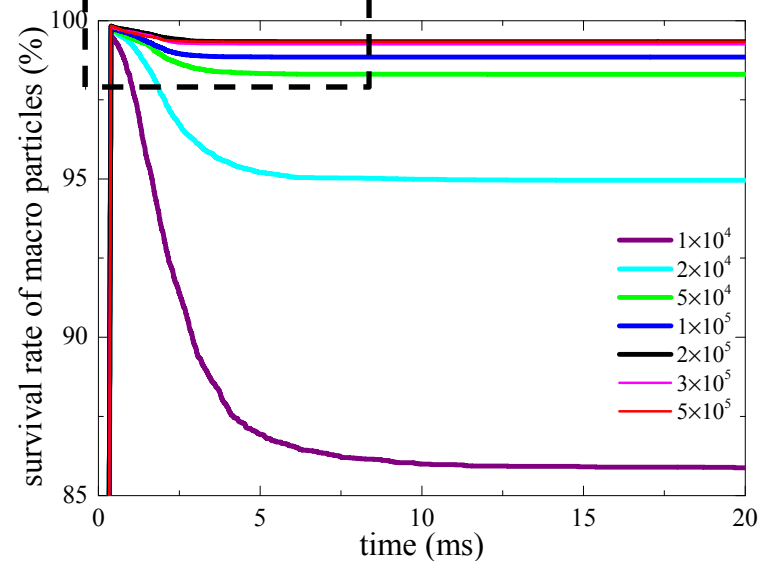
# An ORBIT instance



- To achieve high computing precision and fast execution speed of the program, **model parameters of an ORBIT instance** are studied in detail, including **the number of macro particles** and **the number of turns for the beam being turned around the ring**.



- In the RCS, the proton beam is accumulated through an anti-correlated painting scheme within 200 turns, and accelerated to 1.6 GeV in about 20 000 turns. It takes about **20 ms per cycle**.



- The stability of simulation results increases with the number of macro particles.

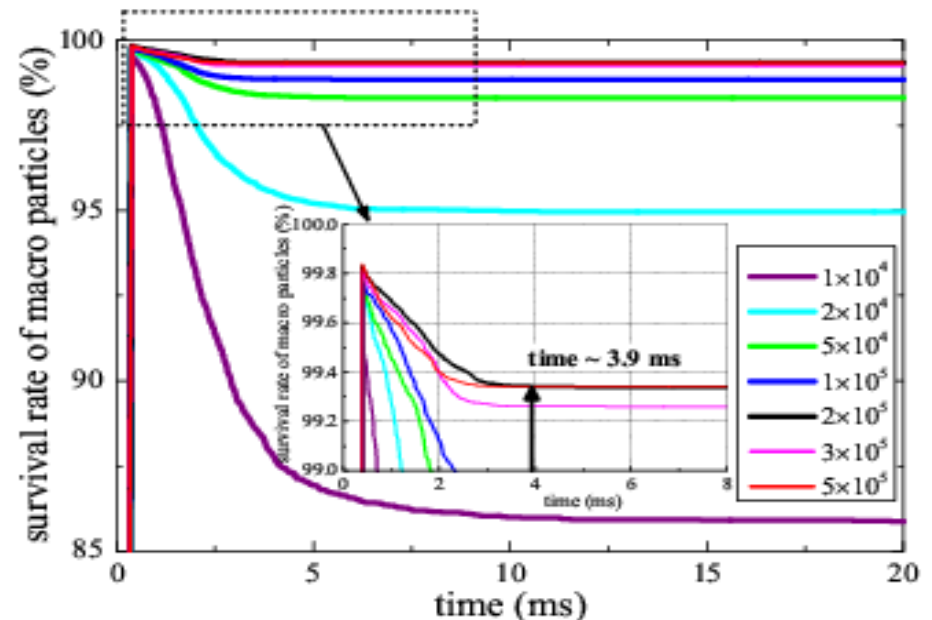
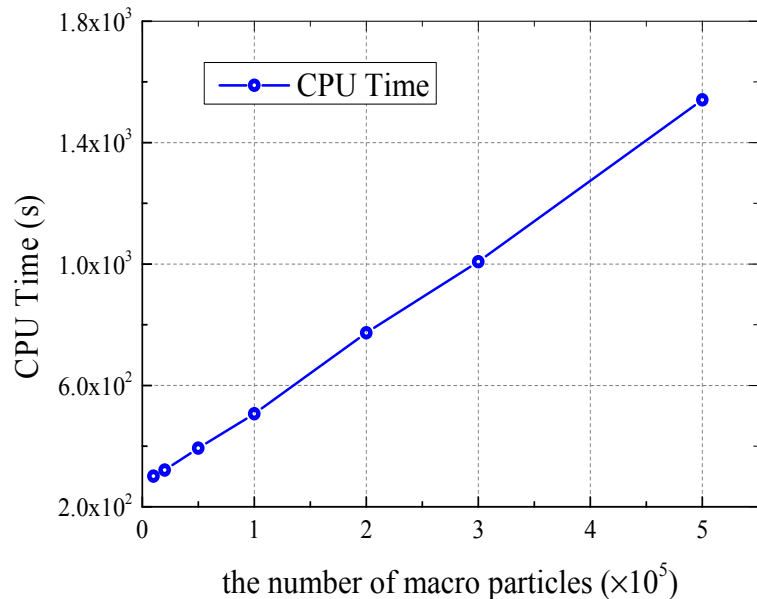
- The simulation time is related to the number of turns for the acceleration.

- With  $2 \times 10^5$  macro particles, the beam loss mainly occurs within the first 3.9 ms, which corresponds to 2200 turns for beam tracking.

# An ORBIT instance



- To achieve high computing precision and fast execution speed of the program, **model parameters of an ORBIT instance** are studied in detail, including **the number of macro particles** and **the number of turns for the beam being turned around the ring**.
- The stability of simulation results increases with the number of macro particles.
- The number of macro particles directly affects the simulation time.



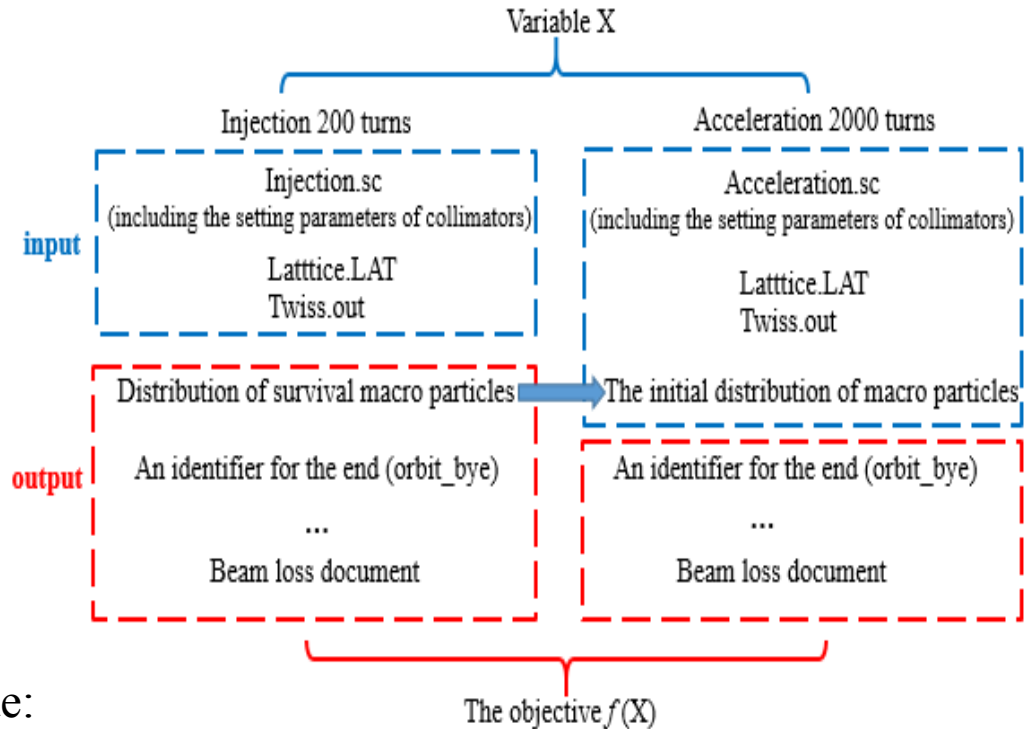
- Apparently, the CPU time increases approximately linearly with the number of macro particles with the same computing resources at IHEP.

# Implementation of the simulation code



- An ORBIT instance:

Model parameters / unit	Value
number of real protons / $\times 10^{12}$	7.8
number of macro particles / $\times 10^5$	2
number of injection turns	200
number of acceleration turns	2000
beam tracking, turns / time	2200 turns / 3.9 ms



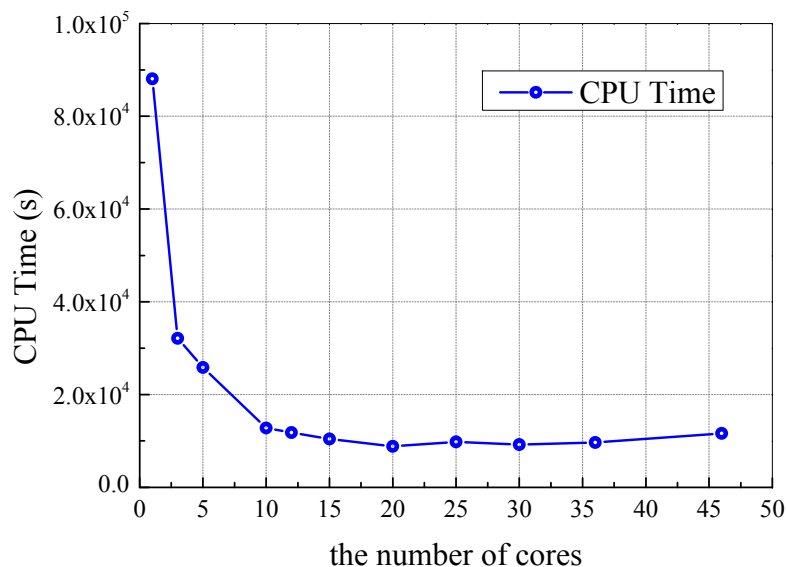
- Implementation of the simulation code:

- A complete ORBIT instance for CSNS/RCS consists of two parts, the injection and acceleration.
- Macro particles are generated and put into the injection simulation. The output of the injection simulation is then used as input for the acceleration simulation.
- To accommodate the searching process of the algorithm **in an automatic way**, a code is first **written in MATLAB** to create and submit an ORBIT instance automatically.

# The computing resources and the limitation



- A cluster in a Gigabit network environment at IHEP
  - Each node of the cluster has 2 CPUs;
  - Each CPU has 6 cores.



- Compared the CPU time of running an ORBIT instance with different numbers of cores, and the distributions of the cores over different nodes are listed in the Table:
  - The CPU time decreases obviously with the increase of the number of cores on the same node.
  - When the cores belong to different nodes, an increase of the number of cores would not lead to a proportionate decrease in the CPU time of running an ORBIT instance.
  - The communication loss between nodes is not ignorable.

cores	1	3	5	10	12	15	20	25	30	36	46
nodes	1	1	1	1	1	2	2	3	3	4	5

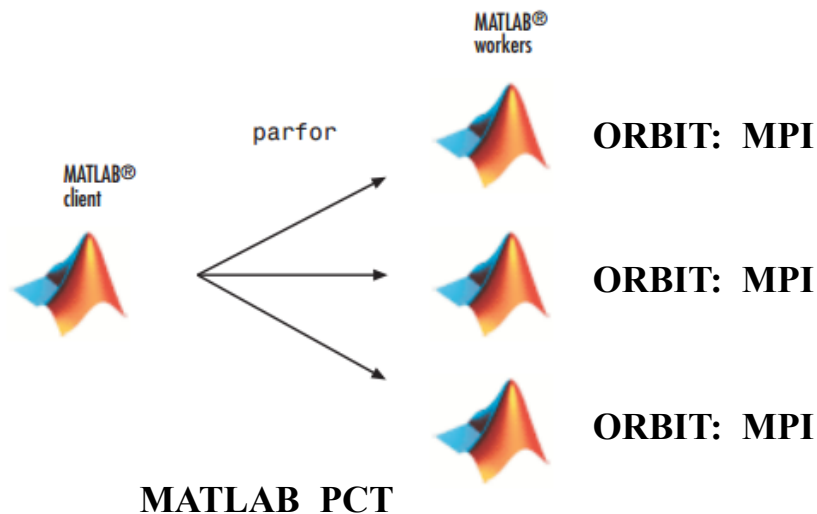
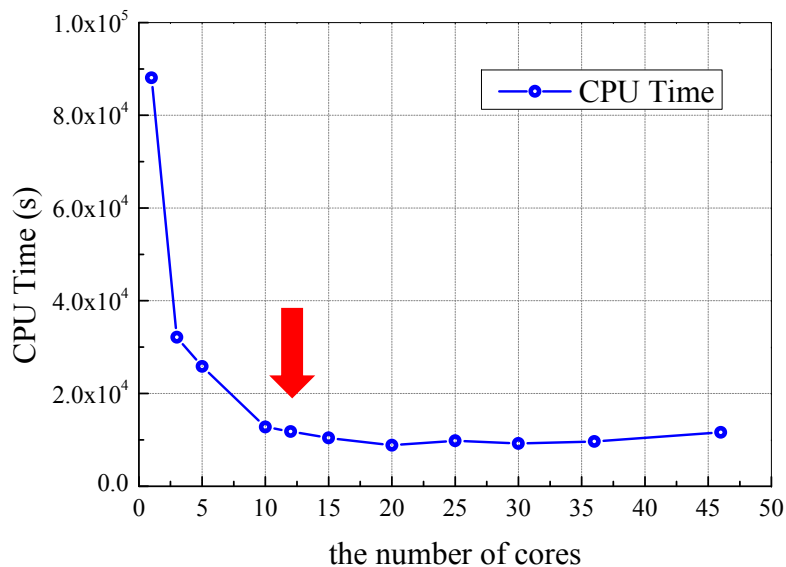


# Parallel computing of ORBIT instances



- A cluster in a Gigabit network environment at IHEP

- Each node of the cluster has 2 CPUs;
- Each CPU has 6 cores.



- To make better use of nodes in the cluster and speed up computation of the RCDS method:

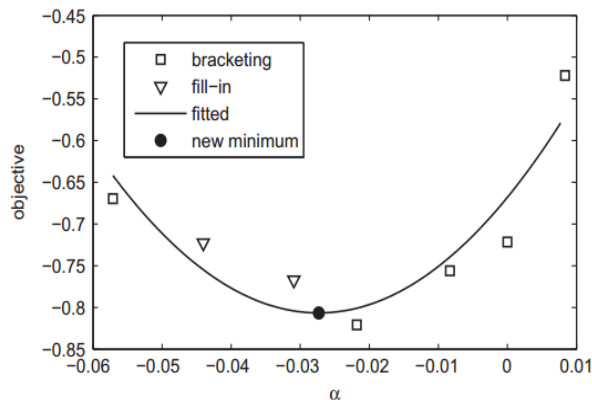
- 12 cores **inside the same node** are used to run an instance **with the MPI parallel computing**.
- A function is programmed within the RCDS algorithm to run several instances **on different nodes simultaneously** with the MATLAB parallel computing toolbox.

# Parallel computing of ORBIT instances

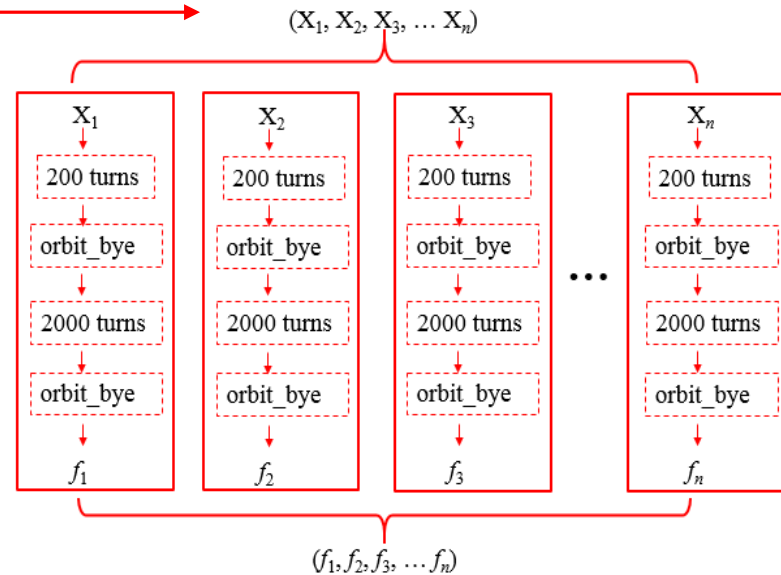


## Parallelorbit.m

➤  $x_0$ , step, range



【X. Huang, J. Corbett, J. Safranek, J. Wu, NIM A 726 (2013) 77-83】



- For a certain direction, the starting point, the step size and the range for the variables can be determined before the line optimizer is executed. So the candidate values of variables are calculated to give out a variables vector set  $(X_1, X_2, \dots, X_n)$ , and the corresponding ORBIT instances are created.
- The function we defined in MATLAB is called to distribute these ORBIT instances over  $n$  nodes individually with PCT, and then run each ORBIT instance inside a separate node with MPI.
- Finally, the corresponding objective values across all the separate nodes are obtained to form a library.
- Set up a candidate function library,  $\{(X_1, f_1), (X_2, f_2), \dots, (X_n, f_n)\}$ , for the present direction
- In this method, a group of candidate evaluations of the objective are obtained in a period of time during which the original method can get only one evaluation.



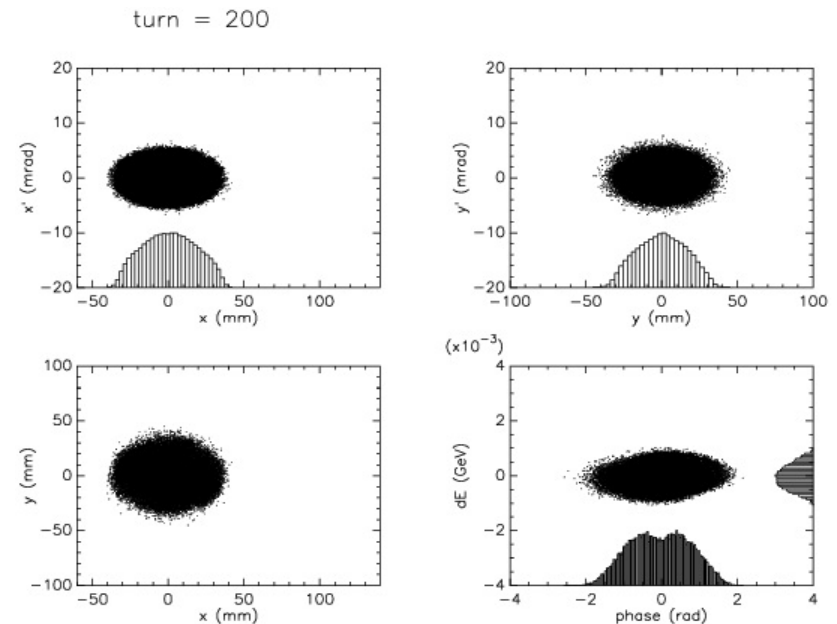
1. Introduction
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  - 3.2 The initial values of variables
  - 3.3 The noise level
  - 3.4 Optimization result
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# The beam distribution



- In order to confirm the optimization of the performance of the collimation system, we **fix the particle distribution for the input of the acceleration process.**
- **A realistic distribution of macro particles** is obtained with:
  - The acceptances of secondary collimators being set to  $500 \pi\text{mm}\cdot\text{mrad}$ .
  - An anti-correlated painting scheme.
- The horizontal 99% emittance of **the beam distribution** is  $193 \pi\text{mm}\cdot\text{mrad}$  and the vertical is  $219 \pi\text{mm}\cdot\text{mrad}$ .

parameters	values
$\varepsilon_{x, \text{rms}} (\pi\text{mm}\cdot\text{mrad})$	0.639
$\varepsilon_{y, \text{rms}} (\pi\text{mm}\cdot\text{mrad})$	0.619
$\alpha_x$	0.003
$\alpha_y$	-0.102
$\beta_x (\text{m})$	1.631
$\beta_y (\text{m})$	1.637
$E_{\text{mean}} (\text{MeV})$	80.4
$\sigma_E (\text{MeV})$	0.15



# The configuration parameters



## The initial values of variables

- The initial acceptances of secondary collimators are set to  $420 \pi\text{mm}\cdot\text{mrad}$ .
- Their ranges are tuned from  $370 \pi\text{mm}\cdot\text{mrad}$  to  $500 \pi\text{mm}\cdot\text{mrad}$  due to the acceptance of the primary collimator and the transverse acceptance of the ring.

## The noise level

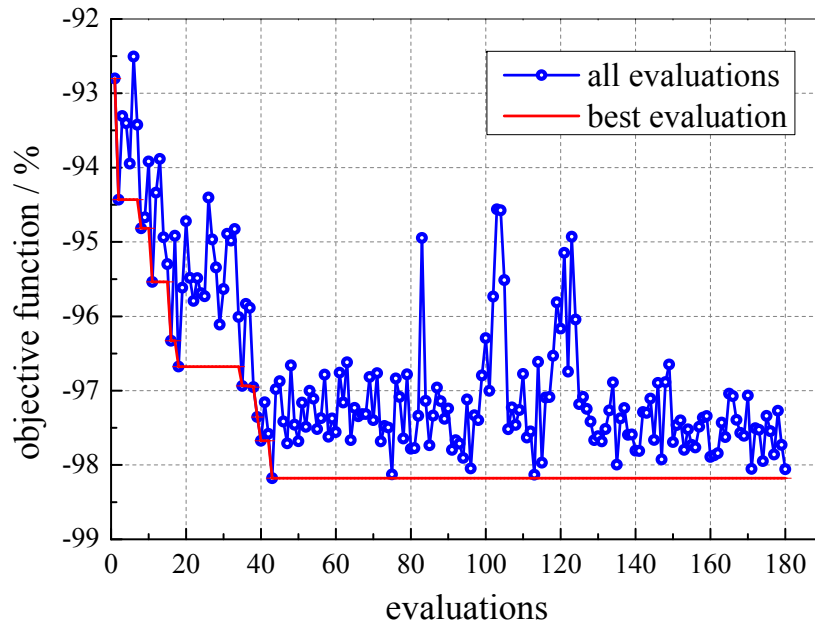
- By running an instance with the beam being accelerated for 2000 turns repeatedly, the noise level of the cleaning efficiency is calculated to be 0.7%, and this value is used as the noise of the objective during the optimization.

$$\sigma_{\text{noise}} = 0.7 \%$$

# Optimization result



- Having configured the parameters of RCDS, the simulation is performed:



Element	variables	value / mm	Acceptance / $\pi\text{mm}\cdot\text{mrad}$	$\lambda_{\text{co}} / \%$
CS1	$c_1$	56.1	371	29.58
	$c_2$	48.3	470	
CS2	$c_3$	60.9	372	58.83
	$c_4$	55.8	419	
CS3	$c_5$	59.5	420	5.08
	$c_6$	56.6	420	
CS4	$c_7$	50.6	420	2.65
	$c_8$	48.2	420	

➤ The objective: -92.8% → -98.2%

➤ The objective is optimized automatically within 180 evaluations. The simulation time is lower than the time of running one ORBIT instance 180 times.

# Optimization result



- A comparison of the parameters reflecting the performance of the collimation system between **the initial state** and **the optimal result**:

parameters	$\eta_{\text{system}}$ / %	$\lambda_{\text{un}}$ / $10^{-4}$	$\eta_{\text{co}}$ / %	$\lambda_{\text{total}}$ / %	$\epsilon_x$ / $\pi\text{mm}\cdot\text{mrad}$	$\epsilon_y$ / $\pi\text{mm}\cdot\text{mrad}$
initial state	92.8	4.9	91.9	0.7	193	219
optimal result	98.2	1.7	96.3	0.9	193	215

- **The cleaning efficiency**,  $\eta_{\text{system}}$ , is optimized to 98.2%.
- **The uncontrolled beam loss** of total beam outside the collimation section during the acceleration,  $\lambda_{\text{un}}$ , is lower.
- **The collimation efficiency of collimators**,  $\eta_{\text{co}}$ , is higher.
- **The total beam loss** along the ring,  $\lambda_{\text{total}}$ , is acceptable for shielding, although it is a little higher than that of the initial state.
- **The 99% horizontal (vertical) emittance** of the beam shows a good quality beam.
- Considering even larger beam loss might be caused by various kinds of errors in the actual conditions, it is more important to reduce the uncontrolled beam loss.

# Summary



- We have implemented the RCDS method to optimize the collimation system of CSNS/RCS.
- The uncontrolled beam loss of the total beam during the acceleration can be reduced to  $1.7 \times 10^{-4}$ , which is lower than that obtained by previous optimization.
- As a result, an approach is established to efficiently give an optimal parameter combination of the secondary collimators for the present machine model.



# Thanks for your attention!

